

## AMENDMENTS TO THE CLAIMS

**Claim 1 (Withdrawn)** A combustor comprising:

a pilot nozzle being provided in a center portion of a combustor main body;

a plurality of main nozzles being provided in a surrounding area of the pilot nozzle at regular intervals;

a pilot cone covering a downstream-side tip portion where a fuel of the pilot nozzle flows and having tip portion thereof provided with a tapered portion of inner circumference of a cone which has a tapered shape in a radial pattern toward downstream side; and

pilot swirls being installed so as to be in contact with an inner wall surface of the pilot cone and supporting the pilot nozzle in a center portion of the pilot cone;

wherein, a fuel being injected from fuel injection ports which are provided to an outer circumference of a tip of the pilot nozzle and inject a fuel bumps against an inner wall surface of the tapered portion of inner circumference of a cone from a position being half as long as the tapered portion of inner circumference of a cone to a downstream-side tip.

**Claim 2 (Withdrawn)** A combustor as described in Claim 1:

wherein, when an opening angle of the pilot cone is “θ,” a spray angle “α” of a fuel being injected from the fuel injection ports which are provided to an outer circumference of tip of the pilot nozzle and inject a fuel satisfies:

$$-90^\circ \leq \alpha < -\theta/2, \quad \theta/2 < \alpha \leq 90^\circ$$

**Claim 3 (Withdrawn)** A combustor comprising:

a pilot nozzle being provided in a center portion of a combustor main body;

a plurality of main nozzles being provided in a surrounding area of the pilot nozzle at regular intervals;

a pilot cone covering a downstream-side tip portion where a fuel of the pilot nozzle flows and having tip portion thereof provided with a tapered portion of inner circumference of a cone which has a tapered shape in a radial pattern toward downstream side; and

pilot swirls being installed so as to be in contact with an inner wall surface of the pilot cone and supporting the pilot nozzle in a center portion of the pilot cone:

wherein, when an opening angle of the pilot cone is “ $\theta$ ,” a spray angle of a fuel being injected from fuel injection ports which are provided to an outer circumference of a tip of the pilot nozzle and inject a fuel is “ $\theta/2$ ;” and the fuel is injected in parallel with inclination of the tapered portion of an inner circumference of a cone.

**Claim 4 (Withdrawn)** A combustor as described in Claim 3:

wherein, a distance “c” between a jet flow of a fuel being injected from the pilot cone and a tapered inner wall surface of the tapered portion of inner circumference of a cone is “ $c < 1/2 (B - D)$  ,” wherein a diameter of a downstream-side tip of the pilot cone is “B” and a diameter of a pilot nozzle is “D.”

**Claim 5 (Withdrawn)** A combustor comprising:

a pilot nozzle being provided in a center portion of a combustor main body;

a plurality of main nozzles being provided in a surrounding area of the pilot nozzle at regular intervals;

a pilot cone covering a downstream-side tip portion where a fuel of the pilot nozzle flows and having tip portion thereof provided with a tapered portion of inner circumference of a cone which has a tapered shape in a radial pattern toward downstream side; and

pilot swirls being installed so as to be in contact with an inner wall surface of the pilot cone and supporting the pilot nozzle in a center portion of the pilot cone:

wherein, the said pilot nozzle comprises:

a first fuel supply channel being provided in a center portion of the pilot nozzle and letting a large amount of fuel being supplied to the pilot nozzle pass through;

a second fuel supply channel being provided to a surrounding area of the first fuel supply channel and letting a remaining fuel being supplied to the pilot nozzle pass through;

a cylindrical pilot nozzle cover whose outer wall surface is in contact with an inner wall surface of the pilot swirls, covers a downstream-side tip of the pilot nozzle and induces air passing through an outer circumference of the pilot nozzle to a downstream-side tip of the pilot nozzle;

a first fuel injection pipe which is provided to an outer circumference of a downstream-side tip of the pilot nozzle, penetrates through the pilot nozzle cover from the first fuel supply channel, and injects a fuel being supplied from the first fuel supply channel to an outer circumference of the pilot nozzle cover; and

fuel injection ports which are provided to a position being more upstream side than the

first fuel injection pipe on an outer circumference of the pilot nozzle and are connected to the second fuel supply channel so as to inject a fuel being supplied from the second fuel supply channel to an area consisting of the pilot nozzle cover and the pilot nozzle.

**Claim 6 (Withdrawn)** A combustor as described in Claim 5:

wherein, the pilot nozzle cover comprises:

a first cylinder cover being in contact with inner wall surfaces of the pilot swirls and covering a downstream-side tip of the pilot nozzle from a position which is more upstream side than a position of the pilot swirls to a position which is more downstream side than a position of the fuel injection ports; and

a second cylinder cover being provided to a position which overlaps the first cylinder cover and is between the pilot nozzle and the first cylinder cover, wherein the first fuel injection pipe penetrates through.

**Claim 7 (Withdrawn)** A combustor as described in Claim 6:

wherein, the second cylinder cover covers a downstream-side tip of the pilot nozzle from a proximity of the fuel injection ports.

**Claim 8 (Withdrawn)** A combustor as described in Claim 6:

wherein, the pilot nozzle is provided with a second fuel injection pipe which penetrates from the fuel injection ports to the second cylinder cover and injects a fuel being supplied from the second fuel supply channel to an area consisting of the first cylinder cover and the second

cylinder cover.

**Claim 9 (Withdrawn)** A combustor as described in Claim 6:

wherein, the first cylinder cover is installed in a manner that a downstream-side tip of the first cylinder cover comes to more upstream side than the first fuel injection pipe.

**Claim 10 (Withdrawn)** A combustor comprising:

a pilot nozzle being provided in a center portion of a combustor main body;

a plurality of main nozzles being provided in a surrounding area of the pilot nozzle at regular intervals;

a pilot cone covering a downstream-side tip portion where a fuel of the pilot nozzle flows and having tip portion thereof provided with a tapered portion of inner circumference of a cone which has a tapered shape in a radial pattern toward downstream side;

pilot swirls being installed so as to be in contact with an inner wall surface of the pilot cone and supporting the pilot nozzle in a center portion of the pilot cone; and

a cylinder being in contact with a downstream-side surface of the pilot swirls and coming to proximity of an outer wall surface of the pilot nozzle being located more downstream side than the pilot swirls, tip portion of the cylinder being provided with a tapered collar being shaped in a radial pattern toward a downstream side

**Claim 11 (Withdrawn)** A combustor as described in Claim 10:

wherein, the collar is provided so that a downstream-side tip of the collar is located more

upstream side than a position where a jet flow of fuel being injected from the pilot nozzle bumps against the collar.

**Claim 12 (Withdrawn)** A combustor as described in Claim 10:

wherein, the collar is provided so that a downstream-side tip of the collar is located more downstream side than a position where a jet flow of fuel being injected from the pilot nozzle bumps against the collar.

**Claim 13 (Withdrawn)** A combustor comprising:

a pilot nozzle being provided to a center portion of a combustor main body;  
and bypass pipes being connected to bypass valves which bypass an air not being used for combustion to a downstream side of a combustor and being provided to a top side of a combustor main body; and

wherein, the pilot nozzle is provided with a plurality of fuel injection ports injecting a fuel being supplied to the pilot nozzle and being installed to an outer circumference of downstream-side tip of the pilot nozzle, at positions excluding a position which is nearest to the bypass pipes.

**Claim 14 (Withdrawn)** A combustor comprising:

a pilot nozzle being provided to a center portion of a combustor main body; and  
connection pipes being provided to a side surface of a combustor main body propagating flames to another combustor; and

wherein, the pilot nozzle is provided with a plurality of fuel injection ports injecting a fuel

being supplied to the pilot nozzle and being installed to an outer circumference of downstream-side tip of the pilot nozzle, at positions excluding a position which is nearest to the connection pipes.

**Claim 15 (Withdrawn)** A combustor comprising:

a pilot nozzle being provided to a center portion of a combustor main body ; and bypass pipes connecting to bypass valves which bypass air being not used for combustion to a downstream side of a combustor and being installed to a top side of a combustor main body; and wherein, the bypass valves are slightly opened.

**Claim 16 (Currently Amended)** A combustor comprising:

a pilot nozzle located at being provided in a center portion of a combustor main body of the combustor;

a plurality of main nozzles located at being provided in a surrounding area of the pilot nozzle, each respective main nozzle of the plurality of main nozzles being spaced apart at a regular interval-intervals;

a pilot cone covering a downstream-side tip portion of the pilot nozzle where a fuel of the pilot nozzle flows; and

main burners covering downstream-side tip portions of the plurality of main nozzles; and wherein, the pilot cone comprises:

a cone having a tapered portion located at-of an inner circumference thereof-of a

eone being formed in, such that the tapered portion of the cone forms a tapered shape stretching outwardly so as to stretch out in a radial pattern toward a downstream side of the combustor; and  
\_\_\_\_\_ a collar portion located at which is provided to an external periphery of a  
downstream-side tip of the tapered portion of the inner circumference of [[a]] the cone, such that  
the collar portion is located on a downstream side of downstream-side tips of the main burners  
and such that the collar portion provides serves as a surface extending being approximately  
vertical to an axial direction of the said pilot nozzles and,

wherein, an angle measured from a line extending in the in axial direction to a line that  
connects connecting an external periphery of a downstream-side tip of the pilot nozzle and with  
an external periphery of the downstream-side tips of the main burners is specified as " $\alpha_x$ ," and  
wherein an opening angle measured at of the tapered portion of the inner circumference of  
[[a]] the cone is specified as " $\theta_s$ " wherein " $0 < \theta < 2\alpha_x$ ," is " $0 \leq \theta < 2\alpha_x$ ,"

**Claim 17 (Currently Amended)** A combustor as described in Claim 16[[:]], wherein the pilot  
cone further comprises:

, an outer circumference of the tapered portion of inner circumference of a cone is-  
provided with a cylinder installed therein, such that a space is formed between the cylinder and  
an outer wall of the tapered portion of the inner circumference of the cone; and  
a ring collar installed at a position in the axial direction of the downstream-side tips of the  
main burners, such that a space is formed between the ring collar and the collar, and such that the  
ring collar provides a surface extending approximately vertical to the axial direction of the pilot

nozzle so as to confront the collar being shaped so as to be along an outer wall of the tapered portion of inner circumference of a cone and the collar portion.

**Claim 18 (Withdrawn)** A combustor as described in Claim 17:

wherein, a downstream-side tip of the cylinder is placed so as to be located in proximity of an external periphery of downstream-side tips of the main burners.

**Claim 19 (Withdrawn)** A combustor as described in Claim 16:

wherein, the collar is installed to more downstream side than downstream-side tips of the main burners for several millimeters.

**Claim 20 (Withdrawn)** A combustor as described in Claim 16:

wherein, the tapered portion of inner circumference of a cone is provided with a cylindrical portion which is extended from a joint portion with the collar portion in the tapered portion of inner circumference of a cone, having a same opening angle “ $\theta$ ” as the tapered portion of inner circumference of a cone.

**Claim 21 (Withdrawn)** A combustor as described in Claim 20 :

wherein, are provided with a first cylinder having a first collar portion which covers from downstream-side tips to upstream side of the main burners on an outer circumference of the tapered portion of inner circumference of a cone, serves as a surface being vertical to axial direction of the pilot cone, and extends toward positions of downstream-side tips of the main

burners on downstream-side tips thereof; and

a second cylinder having a second collar portion which covers from downstream-side tips of the main burners to downstream side on an outer circumference of the tapered portion of inner circumference of a cone, is placed at upstream-side tips thereof, and faces the first collar portion.

**Claim 22 (Withdrawn)** A combustor as described in Claim 20:

wherein, a length being along the cylindrical portion in the cylindrical portion of the tapered portion of inner circumference of a cone is one through three times as long as a vertical length against axial direction of the pilot cone in the collar portion.

**Claim 23 (Withdrawn)** A combustor as described in Claim 16:

wherein, the collar portion is installed more upstream side than downstream-side tips of the main burners; and

wherein, the collar portion has a cylindrical portion in a tapered shape, being along an outer walls of the main burners from an external periphery of the collar portion to downstream-side tips of the main burners.

**Claim 24 (Withdrawn)** A combustor as described in Claim 23:

wherein, the tapered portion of inner circumference of a cone has a cylindrical portion extending from a joint portion with the collar portion in the tapered portion of inner circumference of a cone, having a same opening angle “ $\theta$ ” as the tapered portion of inner

circumference of a cone.

**Claim 25 (Withdrawn)** A combustor as described in Claim 16:

wherein, a plurality of flame-stability-enhancing fuel supply channels are provided so as to pass between the pilot cone and the main burners and supply a fuel to downstream-side of the collar portion being connected to the tapered portion of inner circumference of a cone.

**Claim 26 (Withdrawn)** A combustor as described in Claim 25:

wherein, the flame-stability-enhancing fuel supply channels are installed so as to be located on a line connecting the main nozzles and the pilot nozzle.

**Claim 27 (Withdrawn)** A combustor comprising:

a pilot nozzle being provided in a center portion of a combustor main body;  
a plurality of main nozzles being provided in a surrounding area of the pilot nozzle at regular intervals;  
a pilot cone covering a downstream-side tip portion where a fuel of the pilot nozzle flows; and

main burners which cover downstream-side tips of the main nozzles; and

wherein, the pilot cone comprises:

a tapered portion of inner circumference of a cone being provided to downstream-side tip portion and being formed to be tapered, extending in a radial pattern toward a downstream side; and

a double cylinder , wherein a first cylindrical portion being provided to an outer circumference of the tapered portion of inner circumference of a cone is connected to a second cylindrical portion having a wider opening angle than the first cylindrical portion and being provided to an outer circumference of the first cylindrical portion, at upstream-side tip portions thereof, respectively.

**Claim 28 (Withdrawn)** A combustor as described in Claim 27:

wherein, an outer circumference of the double cylinder is provided with a cylinder being shaped so as to be along an outer wall of the second cylindrical portion.

**Claim 29 (Withdrawn)** A combustor as described in Claim 27:

wherein, a plurality of flame-stability-enhancing fuel supply channels are provided so as to pass between the pilot cone and the main burners and supply a fuel to a hollow being formed by the first and the second cylindrical portions of the double cylinder.

**Claim 30 (Withdrawn)** A combustor as described in Claim 29:

wherein, the flame-stability-enhancing fuel supply channels are installed so as to be located on a line connecting the main nozzles and the pilot nozzle.

**Claim 31 (Withdrawn)** A combustor comprising:

a pilot nozzle being provided in a center portion of a combustor main body;

a plurality of main nozzles being provided in a surrounding area of the pilot nozzle at

regular intervals;

a pilot cone covering a downstream-side tip portion where a fuel of the pilot nozzle flows;

and

main burners which cover downstream-side tips of the main nozzles; and

wherein, the pilot cone comprises:

a tapered portion of inner circumference of a cone being provided to downstream-side tip portion and being formed to be tapered, extending in a radial pattern to proximity of downstream-side tips of the main burners,

a first cylindrical portion protruding from an external periphery of downstream-side tip of the tapered portion of inner circumference of a cone to centers of the main burners;

a second cylindrical portion protruding from an external periphery of downstream-side tip of the tapered portion of inner circumference of a cone to centers of the main burners; and

a cylinder being formed so as to be along an outer wall of the tapered portion of inner circumference of a cone and have downstream-side tip thereof be in contact with downstream-side tips of the main burners.

**Claim 32 (Withdrawn)** A combustor as described in Claim 31:

wherein, a length of the second cylindrical portion is approximately same as a length of the first cylindrical portion.

**Claim 33 (Withdrawn)** A combustor as described in Claim 31:

wherein, the second cylindrical portion is formed so as to be closed toward downstream

side within a range between 0 degree and 60 degrees against axial direction of the pilot cone.

**Claim 34 (Withdrawn)** A combustor as described in Claim 31:

wherein, a plurality of flame-stability-enhancing fuel supply channels are provided so as to pass between the pilot cone and the main burners and supply a fuel to an area being surrounded by the first and the second cylindrical portions.

**Claim 35 (Withdrawn)** Combustor as described in Claim 34:

wherein, the flame-stability-enhancing fuel supply channels are installed so as to be located on a line connecting the main nozzles and the pilot nozzle.

**Claim 36 (Withdrawn)** A combustor as described in Claim 17:

wherein, a plurality of flame-stability-enhancing fuel supply channels are provided so as to pass between the pilot cone and the main burners and supply a fuel to downstream-side of the collar portion being connected to the tapered portion of inner circumference of a cone.

**Claim 37 (Withdrawn)** A combustor as described in Claim 18:

wherein, a plurality of flame-stability-enhancing fuel supply channels are provided so as to pass between the pilot cone and the main burners and supply a fuel to downstream-side of the collar portion being connected to the tapered portion of inner circumference of a cone.

**Claim 38 (Withdrawn)** A combustor as described in Claim 19:

wherein, a plurality of flame-stability-enhancing fuel supply channels are provided so as to pass between the pilot cone and the main burners and supply a fuel to downstream-side of the collar portion being connected to the tapered portion of inner circumference of a cone.

**Claim 39 (Withdrawn)** A combustor as described in Claim 20:

wherein, a plurality of flame-stability-enhancing fuel supply channels are provided so as to pass between the pilot cone and the main burners and supply a fuel to downstream-side of the collar portion being connected to the tapered portion of inner circumference of a cone.

**Claim 40 (Withdrawn)** A combustor as described in Claim 21:

wherein, a plurality of flame-stability-enhancing fuel supply channels are provided so as to pass between the pilot cone and the main burners and supply a fuel to downstream-side of the collar portion being connected to the tapered portion of inner circumference of a cone.

**Claim 41 (Withdrawn)** A combustor as described in Claim 22:

wherein, a plurality of flame-stability-enhancing fuel supply channels are provided so as to pass between the pilot cone and the main burners and supply a fuel to downstream-side of the collar portion being connected to the tapered portion of inner circumference of a cone.

**Claim 42 (Withdrawn)** A combustor as described in Claim 23:

wherein, a plurality of flame-stability-enhancing fuel supply channels are provided so as to pass between the pilot cone and the main burners and supply a fuel to downstream-side of the

collar portion being connected to the tapered portion of inner circumference of a cone.

**Claim 43 (Withdrawn)** A combustor as described in Claim 24:

wherein, a plurality of flame-stability-enhancing fuel supply channels are provided so as to pass between the pilot cone and the main burners and supply a fuel to downstream-side of the collar portion being connected to the tapered portion of inner circumference of a cone.

**Claim 44 (Withdrawn)** A combustor as described in Claim 36:

wherein, the flame-stability-enhancing fuel supply channels are installed so as to be located on a line connecting the main nozzles and the pilot nozzle.

**Claim 45 (Withdrawn)** A combustor as described in Claim 37:

wherein, the flame-stability-enhancing fuel supply channels are installed so as to be located on a line connecting the main nozzles and the pilot nozzle.

**Claim 46 (Withdrawn)** A combustor as described in Claim 38:

wherein, the flame-stability-enhancing fuel supply channels are installed so as to be located on a line connecting the main nozzles and the pilot nozzle.

**Claim 47 (Withdrawn)** A combustor as described in Claim 39:

wherein, the flame-stability-enhancing fuel supply channels are installed so as to be located on a line connecting the main nozzles and the pilot nozzle.

**Claim 48 (Withdrawn)** A combustor as described in Claim 40:

wherein, the flame-stability-enhancing fuel supply channels are installed so as to be located on a line connecting the main nozzles and the pilot nozzle.

**Claim 49 (Withdrawn)** A combustor as described in Claim 41:

wherein, the flame-stability-enhancing fuel supply channels are installed so as to be located on a line connecting the main nozzles and the pilot nozzle.

**Claim 50 (Withdrawn)** A combustor as described in Claim 42:

wherein, the flame-stability-enhancing fuel supply channels are installed so as to be located on a line connecting the main nozzles and the pilot nozzle.

**Claim 51 (Withdrawn)** A combustor as described in Claim 43:

wherein, the flame-stability-enhancing fuel supply channels are installed so as to be located on a line connecting the main nozzles and the pilot nozzle.

**Claim 52 (Withdrawn)** A combustor as described in Claim 28:

wherein, a plurality of flame-stability-enhancing fuel supply channels are provided so as to pass between the pilot cone and the main burners and supply a fuel to a hollow being formed by the first and the second cylindrical portions of the double cylinder.

**Claim 53 (Withdrawn)** A combustor as described in Claim 52:

wherein, the flame-stability-enhancing fuel supply channels are installed so as to be located on a line connecting the main nozzles and the pilot nozzle.

**Claim 54 (Withdrawn)** A combustor as described in Claim 32:

wherein, a plurality of flame-stability-enhancing fuel supply channels are provided so as to pass between the pilot cone and the main burners and supply a fuel to an area being surrounded by the first and the second cylindrical portions.

**Claim 55 (Withdrawn)** A combustor as described in Claim 33:

wherein, a plurality of flame-stability-enhancing fuel supply channels are provided so as to pass between the pilot cone and the main burners and supply a fuel to an area being surrounded by the first and the second cylindrical portions.

**Claim 56 (Withdrawn)** Combustor as described in Claim 54:

wherein, the flame-stability-enhancing fuel supply channels are installed so as to be located on a line connecting the main nozzles and the pilot nozzle.

**Claim 57 (Withdrawn)** Combustor as described in Claim 55:

wherein, the flame-stability-enhancing fuel supply channels are installed so as to be located on a line connecting the main nozzles and the pilot nozzle.